

Controlled reattachment in separated flows: a variational approach to recirculation length reduction

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A variational technique is used to derive analytical expressions for the sensitivity of recirculation length to steady forcing in separated flows. Linear sensitivity analysis is applied to the two-dimensional steady flow past a circular cylinder for Reynolds numbers $40 \leq Re \leq 120$, both in the subcritical and supercritical regimes. Regions which are the most sensitive to volume forcing and wall blowing/suction are identified. Control configurations which reduce the recirculation length are designed based on the sensitivity information, in particular small cylinders used as control devices in the wake of the main cylinder, and fluid suction at the cylinder wall. Validation against full non-linear Navier-Stokes calculations shows excellent agreement for small-amplitude control. The linear stability properties of the controlled flow are systematically investigated. At moderate Reynolds numbers, we observe that regions where control reduces the recirculation length correspond to regions where it has a stabilising effect on the most unstable global mode associated to vortex shedding, while this property does not hold any more at larger Reynolds numbers.

Key words: flow control, wakes/jets, separated flows

1. Introduction

Separation occurs in flow configurations with abrupt geometry changes or strong adverse pressure gradients. In practical engineering applications, separation is generally associated with low-frequency fluctuations which can have undesirable effects, e.g. deterioration of vehicle performance, fatigue of mechanical structures, etc. The control of separated flows is therefore an active research area. Part of the ongoing research work focuses on the laminar regime, where separated flows are steady at low Reynolds number and become unsteady above a threshold value. In this regime, stability theory can help design control strategies by providing insight into the physical phenomena involved in the transition to unsteadiness through, for example, the transient growth of particular initial perturbations, or the bifurcation of unsteady eigenmodes. Examples of separated flows commonly studied as archetypical configurations because of their fundamental interest include bluff bodies (e.g. square and circular cylinders), backward-facing steps, bumps, stenotic geometries, and pressure-induced separations over flat plates.

Sensitivity analysis uses a variational approach to calculate efficiently the linear sensitivity of some quantity to a modification of the flow or to a given actuation, thus

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suppressing the need to resort to exhaustive parametric studies. Hill (1992) applied sensitivity analysis to the flow past a cylinder and computed the sensitivity of the most unstable growth rate to passive control by means of a second smaller cylinder, and successfully reproduced most sensitive regions previously identified experimentally by Strykowski & Sreenivasan (1990). Since then, sensitivity analysis gained popularity and was applied to evaluate sensitivity to flow modification or to passive control in various flows, both in local and global frameworks. For example, Corbett & Bottaro (2001) designed a control strategy based on such a variational technique in order to reduce optimal transient growth in boundary layers. This was achieved by computing the sensitivity of an objective function involving energy, which was then iteratively minimized. Such quadratic cost functionals are very often employed in control theory, but sensitivity analysis can be applied to non-quadratic quantities as well. Bewley *et al.* (2001) minimized several kinds of cost functionals and successfully relaminarized the turbulent flow in a plane channel using wall transpiration. Bottaro *et al.* (2003) used a variational approach to compute the sensitivity of eigenvalues to base flow modification in the parallel plane Couette flow, as well as the most destabilising modification. Marquet *et al.* (2008) studied the sensitivity of the cylinder flow leading eigenvalue to base flow modification and to steady forcing in the bulk and, again, reproduced the regions of Strykowski & Sreenivasan (1990). Meliga *et al.* (2010) managed to control the first oscillating eigenmode in the compressible flow past a slender axisymmetric body by considering its sensitivity to steady forcing, both in the bulk (with mass, momentum or energy sources) and at the wall (with blowing/suction or heating). Recently, Brandt *et al.* (2011) also applied sensitivity analysis to evaluate the effect of steady control on noise amplification (maximal energy amplification under harmonic forcing in steady-state regime) in a globally stable flat-plate boundary layer.

In the present study, sensitivity analysis is applied to another quantity of interest in separated flows: the length of the recirculation region l_c . Many authors observed that in separated flows the recirculation length increases with Re (below the onset of instability): circular cylinder (experimental study by Taneda (1956), numerical study by Giannetti & Luchini (2007)), backward-facing step (experimental study by Acrivos *et al.* (1968), numerical study by Barkley *et al.* (2002)), wall-mounted bump (numerical study by Marquillie & Ehrenstein (2003), experimental study by Passaggia *et al.* (2012)), etc. As the recirculation region gets longer, both maximal backward flow and maximal shear increase. From a local stability viewpoint, this tends to destabilise the flow. In addition, since the shear layer elongates, incoming or developing perturbations are amplified over a longer distance while advected downstream, and any region of absolute instability is increased in length too. When the flow becomes unstable and unsteady, as is the case for the cylinder flow above threshold ($Re > Re_c$), the mean recirculation length decreases (Nishioka & Sato 1978). This is interpreted as the result of a mean flow correction, and the decrease in the mean value of l_c naturally appears as a characteristic global order parameter of the bifurcation (Zielinska *et al.* 1997).

The recirculation length therefore appears as a relevant macroscopic scalar parameter to characterize separated flows. This motivates the design of control strategies which directly target l_c , rather than eigenmode growth rates, transient growth, or noise amplification. In other words, we propose control strategies which do not focus on the fate of perturbations but act upon a feature of the base flow itself.

We choose to design control configurations based on the steady-state base flow, and consider both subcritical and supercritical Reynolds numbers, $40 \leq Re \leq 120$. In the supercritical regime $Re > Re_c$, the uncontrolled steady-state base flow is linearly unstable and the actual flow observed in experiments or numerical simulations is unsteady; but the

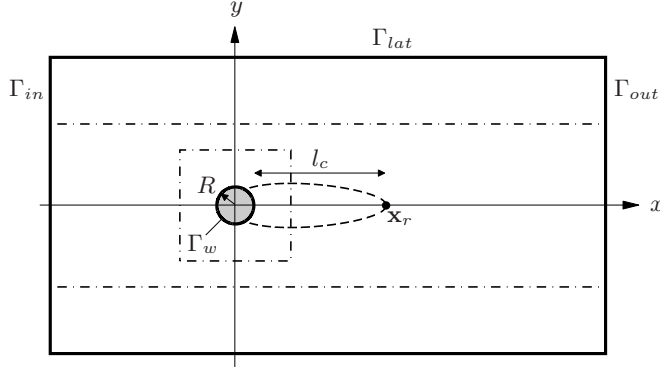


FIGURE 1. Schematic of the problem geometry and computational domain.

sensitivity of the steady-state recirculation length is of interest since reducing l_c might restabilise the flow. The stability of the controlled flow will be assessed systematically to determine when this approach is relevant.

This paper is organized as follows. Section 2 details the problem formulation and numerical methods. In particular, analytical expressions are derived for sensitivity of recirculation length to base flow modifications and to steady control, both volume forcing and wall blowing/suction. Results are presented in section 3: regions sensitive to forcing are identified, and several control configurations which allow to reduce l_c are selected to illustrate the method and to validate the sensitivity analysis against fully non-linear simulations. The linear stability properties of these controlled flows are investigated and discussed in section 4. Conclusions are drawn in section 5.

2. Problem formulation and numerical methods

The sensitivity of the recirculation length in a two-dimensional incompressible cylinder flow is investigated. A cylinder of radius R is located in a uniform flow. The fluid motion is described by the velocity field $\mathbf{U} = (U, V)^T$ of components U and V in the streamwise and cross-stream directions x and y , and the pressure field P . The state vector $\mathbf{Q} = (\mathbf{U}, P)^T$ is solution of the two-dimensional incompressible Navier–Stokes equations

$$\nabla \cdot \mathbf{U} = 0, \quad \partial_t \mathbf{U} + \nabla \mathbf{U} \cdot \mathbf{U} + \nabla P - Re^{-1} \nabla^2 \mathbf{U} = \mathbf{F} \quad (2.1)$$

where $Re = U_\infty D / \nu$ is the Reynolds number based on the cylinder diameter $D = 2R$, the freestream velocity U_∞ and the fluid kinematic viscosity ν , and \mathbf{F} is a steady volume forcing in the bulk. The following boundary conditions are prescribed: uniform velocity profile $\mathbf{U}_b = (U_\infty, 0)^T$ at the inlet Γ_{in} , symmetry condition $\partial_y U_b = 0, V_b = 0$ on lateral boundaries Γ_{lat} , outflow condition $-P_b \mathbf{n} + Re^{-1} \nabla \mathbf{U}_b \cdot \mathbf{n} = \mathbf{0}$ on Γ_{out} , where \mathbf{n} is the normal unit vector oriented outward the domain, blowing/suction $\mathbf{U}_b = \mathbf{U}_w$ on the wall control region Γ_c , and no-slip condition $\mathbf{U}_b = \mathbf{0}$ on the remaining cylinder wall region $\Gamma_w \setminus \Gamma_c$. In this paper attention is restricted to steady flows $\mathbf{Q}_b(x, y)$ which satisfy:

$$\nabla \cdot \mathbf{U}_b = 0, \quad \nabla \mathbf{U}_b \cdot \mathbf{U}_b + \nabla P_b - Re^{-1} \nabla^2 \mathbf{U}_b = \mathbf{F}. \quad (2.2)$$

2.1. Sensitivity of recirculation length

Assuming the flow is symmetric with respect to the symmetry axis $y = 0$, the recirculation length is defined as the distance from the cylinder wall rearmost point $(R, 0)$ to the

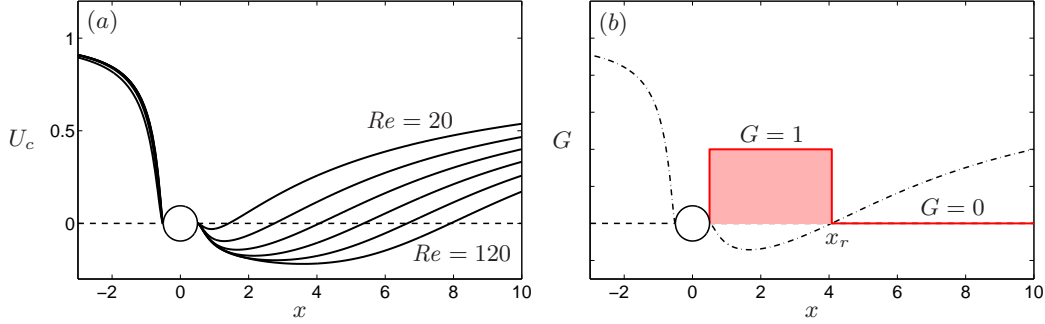


FIGURE 2. (a) Streamwise velocity along the symmetry axis at $Re = 20, 40 \dots 120$. (b) Integrant in the definition of the recirculation length (2.5), illustrated at $Re = 60$.

reattachment point $\mathbf{x}_r = (x_r, 0)$ as shown in figure 1:

$$l_c = x_r - R. \quad (2.3)$$

The reattachment point is characterized by zero streamwise velocity, $U(x_r, 0) = 0$, and can therefore be computed with a bisection method on $U_c(x) = U(x, 0)$ along the symmetry axis.

Throughout this study, only flow modifications and forcings which are symmetric with respect to the symmetry axis will be considered; they result in symmetric flows, thus ensuring that the recirculation length (2.3) is well defined.

2.1.1. Sensitivity to base flow modification

Considering a small modification of the base flow $\delta\mathbf{Q}$, the variation of the recirculation length is expressed at first order as

$$\delta l_c = (\nabla_{\mathbf{Q}} l_c | \delta\mathbf{Q}) \quad (2.4)$$

where $\nabla_{\mathbf{Q}} l_c = (\nabla_{\mathbf{U}} l_c, \nabla_P l_c)^T$ is the sensitivity to base flow modification, $(\mathbf{a} | \mathbf{b}) = \int_{\Omega} \bar{\mathbf{a}} \cdot \mathbf{b} \, d\Omega$ denotes the two-dimensional inner product for real or complex fields, and the overbar stands for complex conjugate.

To allow for the calculation of this sensitivity, the recirculation length is rewritten as

$$l_c = \int_R^\infty H(-U_c(x)) \, dx = \int_R^\infty G(x) \, dx, \quad (2.5)$$

where $U_c(x) = U(x, 0)$ is the streamwise velocity on the symmetry axis and H is the Heaviside step function defined as $H(\alpha) = 0$ for $\alpha < 0$ and $H(\alpha) = 1$ for $\alpha > 0$. As illustrated in figure 2, the integrand is equal to 1 in the recirculation region where $U_c(x) < 0$, and is equal to 0 downstream, therefore integrating along x from the rear stagnation point gives the recirculation length.

Using the same Lagrangian formalism as Hill (1992), the recirculation length variation due to a base flow modification $\delta\mathbf{Q} = (\delta\mathbf{U}, \delta P)^T$ is obtained as:

$$\delta l_c = \lim_{\epsilon \rightarrow 0} \frac{1}{\epsilon} [l_c(\mathbf{Q} + \epsilon \delta\mathbf{Q}) - l_c(\mathbf{Q})] \quad (2.6a)$$

$$= \lim_{\epsilon \rightarrow 0} \frac{1}{\epsilon} \int_R^\infty [H(-U_c(x) - \epsilon \delta U_c(x)) - H(-U_c(x))] \, dx \quad (2.6b)$$

$$= \int_R^\infty - \left. \frac{dH}{dU} \right|_{U=-U_c(x)} \delta U_c(x) \, dx \quad (2.6c)$$

$$= \int_R^\infty \left(\frac{dU_c}{dx}(x) \right)^{-1} \frac{dG}{dx}(x) \delta U_c(x) dx \quad (2.6d)$$

$$= \int_R^\infty - \left(\frac{dU_c}{dx}(x) \right)^{-1} \delta(x - x_r) \delta U_c(x) dx \quad (2.6e)$$

$$= - \frac{\delta U_c(x_r)}{d_x U_c|_{x=x_r}}, \quad (2.6f)$$

where (2.6d) comes from the differentiation of $G(x) = H(-U_c(x))$ using the chain rule, $\frac{dG}{dx}(x) = - \frac{dH}{dU} \Big|_{U=-U_c(x)} \frac{dU_c}{dx}(x)$, and (2.6e) is the result of $\frac{dG}{dx}(x) = -\delta(x - x_r)$ with $\delta(x)$ the Dirac delta function, since G jumps from 1 to 0 at $x = x_r$. If, for example, the streamwise velocity at the original reattachment point increases, then the recirculation region is shortened: this is understood physically as $U_c(x_r)$ becoming positive and the reattachment point moving upstream, while mathematically $\delta U_c(x_r) > 0$ and (2.6) yield $\delta l_c < 0$ (because $d_x U_c|_{x=x_r} > 0$, see figure 2(a)).

The sensitivity to base flow modification is identified as:

$$\nabla_{\mathbf{U}} l_c = - \frac{1}{d_x U_c|_{x=x_r}} \begin{pmatrix} \delta(x_r, 0) \\ 0 \end{pmatrix}, \quad \nabla_P l_c = 0, \quad (2.7)$$

where $\delta(x, y)$ is the two-dimensional Dirac delta function, such that the inner product (2.4) between the two fields $\nabla_{\mathbf{Q}} l_c$ and $\delta \mathbf{Q}$ is indeed δl_c as expressed by (2.6).

In wall-bounded flows, where reattachment occurs at the wall (for example behind a backward-facing step or a bump, or on a flat plate with adverse pressure gradient), the reattachment point is not characterized by zero streamwise velocity $U(x_r, 0) = 0$, but instead by zero wall shear stress, i.e. $\partial_y U|_{x=x_r, y=0} = 0$ with the wall assumed horizontal and located at $y = 0$ for the sake of simplicity. In this case the sensitivity reads:

$$\nabla_{\mathbf{U}} l_c = - \frac{1}{\partial_{xy} U|_{x=x_r, y=0}} \begin{pmatrix} \delta(x_r, 0) \partial_y \\ 0 \end{pmatrix}, \quad \nabla_P l_c = 0. \quad (2.8)$$

The sensitivity field $\nabla_{\mathbf{Q}} l_c$ in (2.7)-(2.8) is valid for any arbitrary base flow modification $\delta \mathbf{U}$. As noted by Brandt *et al.* (2011), it is possible to derive a restricted sensitivity field for divergence-free base flow modifications. In the case of the cylinder flow, where (2.7) results in a localized Dirac delta function at the reattachment point in the x direction only, this restricted sensitivity field appears to present a dipolar structure.

2.1.2. Sensitivity to forcing

Now the sensitivity of the recirculation length to steady forcing is investigated. One considers a small-amplitude forcing: volume force in the bulk $\delta \mathbf{F}(x, y)$, or blowing/suction $\delta \mathbf{U}_w$ on part of the cylinder wall. The recirculation length variation at first order is

$$\delta l_c = (\nabla_{\mathbf{F}} l_c | \delta \mathbf{F}) + \langle \nabla_{\mathbf{U}_w} l_c | \delta \mathbf{U}_w \rangle \quad (2.9)$$

where $\langle \mathbf{a} | \mathbf{b} \rangle = \int_{\Gamma_c} \bar{\mathbf{a}} \cdot \mathbf{b} d\Gamma$ denotes the one-dimensional inner product on the control boundary. The same Lagrangian formalism as in the previous section yields the sensitivities

$$\nabla_{\mathbf{F}} l_c = \mathbf{U}^\dagger, \quad \nabla_{\mathbf{U}_w} l_c = \mathbf{U}_w^\dagger = -P^\dagger \mathbf{n} - Re^{-1} \nabla \mathbf{U}^\dagger \cdot \mathbf{n}, \quad (2.10)$$

where the so-called adjoint base flow $\mathbf{Q}^\dagger = (\mathbf{U}^\dagger, P^\dagger)^T$ is solution of the linear, non-homogeneous system of equations

$$\nabla \cdot \mathbf{U}^\dagger = 0, \quad -\nabla \mathbf{U}^\dagger \cdot \mathbf{U}_b + \nabla \mathbf{U}_b^T \cdot \mathbf{U}^\dagger - \nabla P^\dagger - Re^{-1} \nabla^2 \mathbf{U}^\dagger = \nabla_{\mathbf{U}} l_c, \quad (2.11)$$

with the boundary conditions $\mathbf{U}^\dagger = \mathbf{0}$ on $\Gamma_{in} \cup \Gamma_w \cup \Gamma_c$, symmetry condition $\partial_y U^\dagger = 0$, $V^\dagger = 0$ on Γ_{lat} , and $P^\dagger \mathbf{n} + Re^{-1} \nabla \mathbf{U}^\dagger \cdot \mathbf{n} + \mathbf{U}^\dagger (\mathbf{U}_b \cdot \mathbf{n}) = \mathbf{0}$ on Γ_{out} .

2.1.3. Effect of a small control cylinder

It is of practical interest to study the effect of a particular kind of passive control on recirculation length and eigenvalues, namely a small control cylinder of diameter $d \ll D$ similar to the one used by Strykowski & Sreenivasan (1990) to suppress vortex shedding in a limited range of Reynolds number above instability threshold. The effect on the base flow of a small control cylinder located at $\mathbf{x}_c = (x_c, y_c)$ is modelled as a steady volume force of same amplitude as the drag force acting on this control cylinder, and of opposite direction:

$$\delta \mathbf{F}(x, y) = -\frac{1}{2} d C_d(x, y) \|\mathbf{U}_b(x, y)\| \mathbf{U}_b(x, y) \delta(x - x_c, y - y_c) \quad (2.12)$$

where C_d is the drag coefficient of the control cylinder and depends on the local Reynolds number $Re_d(x, y) = \|\mathbf{U}_b(x, y)\| d / \nu$. Finally, variations of the recirculation length and eigenvalue (see section 2.2) are calculated from $\delta l_c = (\nabla_{\mathbf{F}} l_c | \delta \mathbf{F})$ and $\delta \sigma = (\nabla_{\mathbf{F}} \sigma | \delta \mathbf{F})$:

$$\delta l_c(x_c, y_c) = -\frac{1}{2} d C_d(x_c, y_c) \|\mathbf{U}_b(x_c, y_c)\| \nabla_{\mathbf{F}} l_c(x_c, y_c) \cdot \mathbf{U}_b(x_c, y_c), \quad (2.13)$$

$$\delta \sigma(x_c, y_c) = -\frac{1}{2} d C_d(x_c, y_c) \|\mathbf{U}_b(x_c, y_c)\| \nabla_{\mathbf{F}} \sigma(x_c, y_c) \cdot \mathbf{U}_b(x_c, y_c). \quad (2.14)$$

For a diameter $d = D/10$ and for the set of Reynolds numbers Re and locations (x_c, y_c) chosen hereafter, the Reynolds number of the control cylinder falls in the range $1 \leq Re_d \leq 15$. The expression of Hill (1992) has been generalized in this range according to $C_d(Re_d) = a + b Re_d^c$, based on a set of experimental data from Finn (1953) and Tritton (1959) and from numerical results obtained by the authors, yielding $a = 0.8558, b = 10.05, c = -0.7004$.

2.2. Linear stability

Writing the flow as the superposition of a steady base flow and time-dependent small perturbations, $\mathbf{Q}(x, y, t) = \mathbf{Q}_b(x, y) + \mathbf{q}'(x, y, t)$, linearising the Navier–Stokes equations (2.1) and using the normal mode expansion $\mathbf{q}'(x, y, t) = \mathbf{q}(x, y) e^{\sigma t}$, with $\sigma = \sigma_r + i\sigma_i$, the following system of equations is obtained:

$$\nabla \cdot \mathbf{u} = 0, \quad \sigma \mathbf{u} + \nabla \mathbf{u} \cdot \mathbf{U}_b + \nabla \mathbf{U}_b \cdot \mathbf{u} + \nabla p - Re^{-1} \nabla^2 \mathbf{u} = \mathbf{0}, \quad (2.15)$$

together with the following boundary conditions: $\mathbf{u} = \mathbf{0}$ on $\Gamma_{in} \cup \Gamma_w \cup \Gamma_c$, symmetry condition $\partial_y u = 0, v = 0$ on Γ_{lat} , and outflow condition $-p \mathbf{n} + Re^{-1} \nabla \mathbf{u} \cdot \mathbf{n} = \mathbf{0}$ on Γ_{out} . Solving this generalized eigenvalue problem yields global modes \mathbf{q} and associated growth rate σ_r and pulsation σ_i .

The sensitivity of an eigenvalue to base flow modification, defined by $\delta \sigma = (\nabla_{\mathbf{U}} \sigma | \delta \mathbf{U})$, can be computed as

$$\nabla_{\mathbf{U}} \sigma = -\nabla \bar{\mathbf{u}}^T \cdot \mathbf{u}^\dagger + \nabla \mathbf{u}^\dagger \cdot \bar{\mathbf{u}}, \quad (2.16)$$

where $\mathbf{q}^\dagger = (\mathbf{u}^\dagger, p^\dagger)^T$ is the adjoint mode associated with σ . The sensitivity to steady forcing, defined by $\delta \sigma = (\nabla_{\mathbf{F}} \sigma | \delta \mathbf{F}) + \langle \nabla_{\mathbf{U}_w} \sigma | \delta \mathbf{U}_w \rangle$, can be computed as

$$\nabla_{\mathbf{F}} \sigma = \mathbf{U}^\dagger, \quad \nabla_{\mathbf{U}_w} \sigma = \mathbf{U}_w^\dagger = -P^\dagger \mathbf{n} - Re^{-1} \nabla \mathbf{U}^\dagger \cdot \mathbf{n} \quad (2.17)$$

where this time the adjoint base flow $(\mathbf{U}^\dagger, P^\dagger)^T$ is solution of the linear system

$$\nabla \cdot \mathbf{U}^\dagger = 0, \quad -\nabla \mathbf{U}^\dagger \cdot \mathbf{U}_b + \nabla \mathbf{U}_b^T \cdot \mathbf{U}^\dagger - \nabla P^\dagger - Re^{-1} \nabla^2 \mathbf{U}^\dagger = \nabla_{\mathbf{U}} \sigma, \quad (2.18)$$

with boundary conditions $\mathbf{U}^\dagger = \mathbf{0}$ on $\Gamma_{in} \cup \Gamma_w \cup \Gamma_c$, $\partial_y U^\dagger = 0, V^\dagger = 0$ on Γ_{lat} , and

	C_D					l_c					Re_c
	20	40	60	100	120	20	40	60	100	120	
Henderson (1995)	2.06	1.54	1.31	1.08	1.01						
Zielinska <i>et al.</i> (1997)						0.94	2.28	3.62	6.30		
Giannetti & Luchini (2007)	2.05	1.54				0.92	2.24	(3.6)	(6.2)	(7.5)	46.7
Sipp & Lebedev (2007)											46.6
Marquet <i>et al.</i> (2008)											46.8
Present study	2.04	1.52	1.30	1.07	1.00	0.92	2.25	3.57	6.14	7.42	46.6

TABLE 1. Drag coefficient C_D and recirculation length l_c for different values of Re , and critical Reynolds number Re_c . Bracketed numbers are estimated from a figure.

$P^\dagger \mathbf{n} + Re^{-1} \nabla \mathbf{U}^\dagger \cdot \mathbf{n} + \mathbf{U}^\dagger (\mathbf{U}_b \cdot \mathbf{n}) + \mathbf{u}^\dagger (\bar{\mathbf{u}} \cdot \mathbf{n}) = \mathbf{0}$ on Γ_{out} . It is possible to relate the sensitivity of a given eigenvalue and the individual sensitivities of its growth rate and pulsation, $\delta \sigma_{r,i} = (\nabla_{\mathbf{U}} \sigma_{r,i} | \delta \mathbf{U})$ according to $\nabla_{\mathbf{U}} \sigma_r = \text{Re}\{\nabla_{\mathbf{U}} \sigma\}$ and $\nabla_{\mathbf{U}} \sigma_i = -\text{Im}\{\nabla_{\mathbf{U}} \sigma\}$. The same relations hold for sensitivity to forcing.

2.3. Numerical method

All calculations are performed using the finite element software *FreeFem++* to generate a two-dimensional triangulation in the computation domain Ω shown in figure 1, of dimensions $-50 \leq x \leq 175$, $-30 \leq y \leq 30$, with the center of cylinder located at $x = 0, y = 0$. Bold lines indicate boundaries. The mesh density increases from the outer boundaries towards the cylinder wall, in successive regions indicated by dash-dotted lines in figure 1. The resulting mesh has 246083 triangular elements. Variational formulations associated to the equations to be solved are spatially discretized using P2 and P1 Taylor-Hood elements for velocity and pressure respectively.

Base flows are computed using an iterative Newton method to solve equations (2.2), convergence being reached when the residual is smaller than 10^{-12} in L^2 norm. The eigenvalue problem (2.15) is solved using an implicitly restarted Arnoldi method. Adjoint base flows involved in the calculation of recirculation length sensitivity and eigenvalue sensitivity are obtained by inverting the simple linear systems (2.11) and (2.18).

Convergence was checked by calculating steady-state base flows with different meshes. Reducing the number of elements by 21%, the recirculation length varied by 0.2% or less over the range of Reynolds numbers $30 \leq Re \leq 120$. Values of drag coefficient and recirculation length over this same Re range are given in table 1 together with results from the literature. Compared to Giannetti & Luchini (2007), the maximum relative difference on l_c was 2%, of the same order as the values they report, while C_D differed by less than 1.7% from values computed by Henderson (1995). From linear stability calculations, the onset of instability characterized by $\sigma_r = 0$ was found to be $Re_c = 46.6$, in good agreement with the values reported in the literature. Also, the frequency σ_i of the most unstable global mode differed by less than 1.2% from results of Giannetti & Luchini (2007).

3. Results

In this section we consider subcritical and supercritical Reynolds numbers, $40 \leq Re \leq 120$, and focus on the steady-state recirculation length. Its sensitivity to flow modification and to control is presented in section 3.1, and examples of control configurations which reduce l_c are detailed in section 3.2. Stability properties are discussed later in section 4.

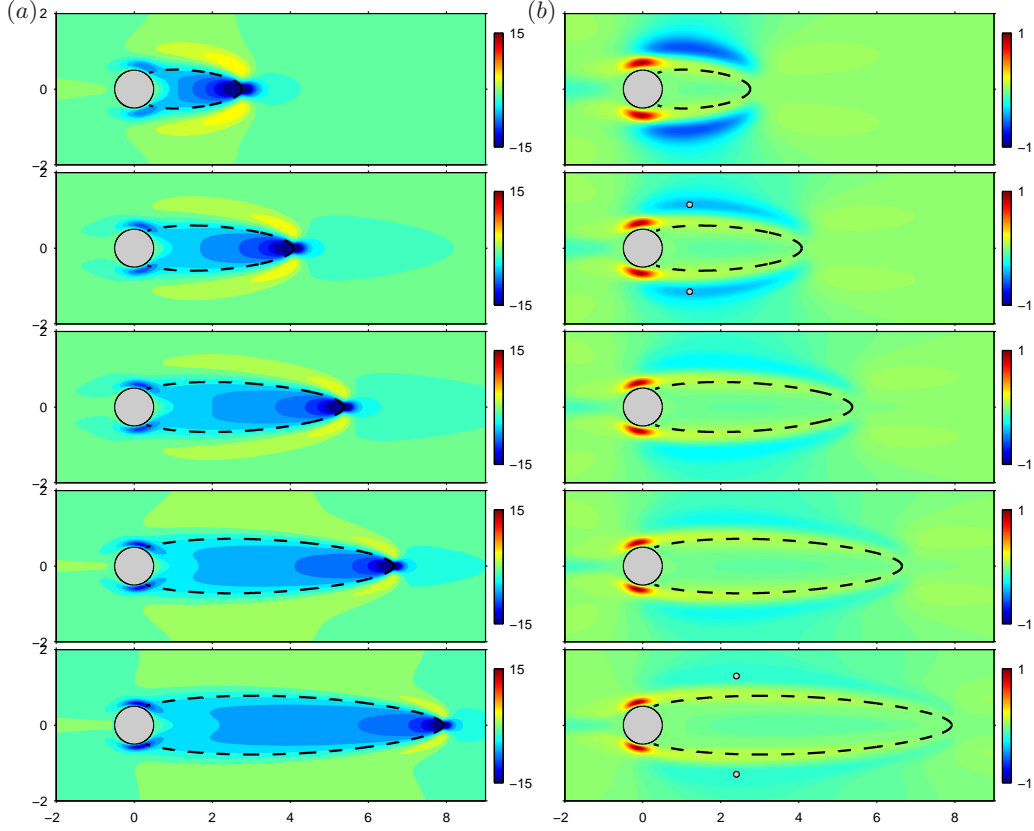


FIGURE 3. (a) Normalized sensitivity of recirculation length to bulk forcing in the streamwise direction, $\nabla_{F_x} l_c / l_c$. (b) Normalized effect of a small control cylinder of diameter $d = 0.10D$ on recirculation length, $\delta l_c / l_c$. From top to bottom: $Re = 40, 60, 80, 100, 120$. The dashed line is the steady-state base flow separatrix. Black circles show the locations of control cylinders for configuration B discussed in section 3.2.

3.1. Sensitivity of recirculation length

Figure 3(a) shows the sensitivity of recirculation length to bulk forcing in the streamwise direction, $\nabla_{F_x} l_c = \nabla_{\mathbf{F}} l_c \cdot \mathbf{e}_x$. By construction, sensitivity analysis predicts that forcing has a large effect on l_c in regions where sensitivity is large. To be more specific, l_c can be reduced by forcing along \mathbf{e}_x in regions of negative sensitivity $\nabla_{F_x} l_c < 0$: in the recirculation region (in particular close to the reattachment point), and near the sides of the cylinder just upstream of the separation points; l_c can also be reduced by forcing along $-\mathbf{e}_x$ in regions of positive sensitivity $\nabla_{F_x} l_c > 0$: at the outer sides of the recirculation region (in particular close to the reattachment point).

As mentioned in section 2.1.3, it is convenient in an experiment to use a small, secondary cylinder as a simple passive control device which produces a force which aligned with the local flow direction and depending non-linearly on the local velocity, as given by (2.12). The effect δl_c of such a control cylinder was computed from $\nabla_{\mathbf{F}} l_c$ according to (2.13). Results for a control cylinder of diameter $d = 0.10D$ are shown in figure 3(b). The recirculation length increases ($\delta l_c > 0$) when a control cylinder is located on the sides of the main cylinder upstream of the separation point, or in the shear layers; it decreases ($\delta l_c < 0$) when a control cylinder is located farther away on the sides of the shear

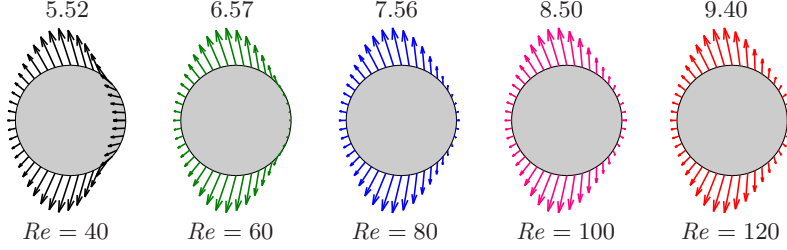


FIGURE 4. Sensitivity of recirculation length to wall actuation $\nabla_{\mathbf{U}_w} l_c$. Flow is from left to right. Numbers correspond to the L^∞ norm of $\nabla_{\mathbf{U}_w} l_c / l_c$.

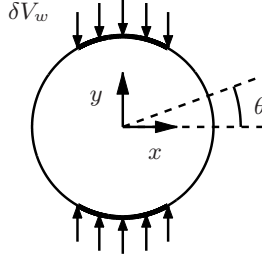


FIGURE 5. Sketch of control configuration W.

layers. The minimal value of $\delta l_c / l_c$ becomes less negative as Re increases, meaning that a control cylinder of constant diameter becomes gradually less effective at reducing l_c . Interestingly, regions associated with l_c reduction correspond qualitatively well to regions where vortex shedding was suppressed in the experiments of Strykowski & Sreenivasan (1990) (figure 20 therein) for the same diameter ratio. This point is further discussed in section 4.

Figure 4 shows the sensitivity of recirculation length to steady wall actuation, $\nabla_{\mathbf{U}_w} l_c$. Since the variation of l_c is given by the inner product $\delta\sigma = \langle \nabla_{\mathbf{U}_w} \sigma \mid \delta \mathbf{U}_w \rangle$, wall control oriented along the arrows increases l_c . In particular, the recirculation length is increased by wall blowing where arrows point towards the fluid domain, and by wall suction where arrows point inside the cylinder. The numbers above each plot correspond to the L^∞ norm of the rescaled sensitivity field $\nabla_{\mathbf{U}_w} l_c / l_c$. This norm increases (roughly linearly) with Re , indicating that the relative control authority is increasing. The shape of the sensitivity field does not vary much with Re , and reveals that the most efficient way to reduce l_c is to use wall suction at the top and bottom sides of the cylinder, in a direction close to wall normal.

3.2. Control of the recirculation length

The sensitivity fields obtained in the previous section can be used to control recirculation length. To illustrate this process, two control configurations predicted to reduce l_c are tested at two representative Reynolds numbers $Re = 60$ and 120 , and for different control amplitudes:

- configuration B (“bulk”; sketched in figure 3): volume forcing with two control cylinders located symmetrically close to the point where l_c reduction is predicted to be maximal, at $\mathbf{x}^* = (x^*, \pm y^*) = (1.2, \pm 1.15)$ at $Re = 60$ and $\mathbf{x}^* = (2.4, \pm 1.3)$ at $Re = 120$;

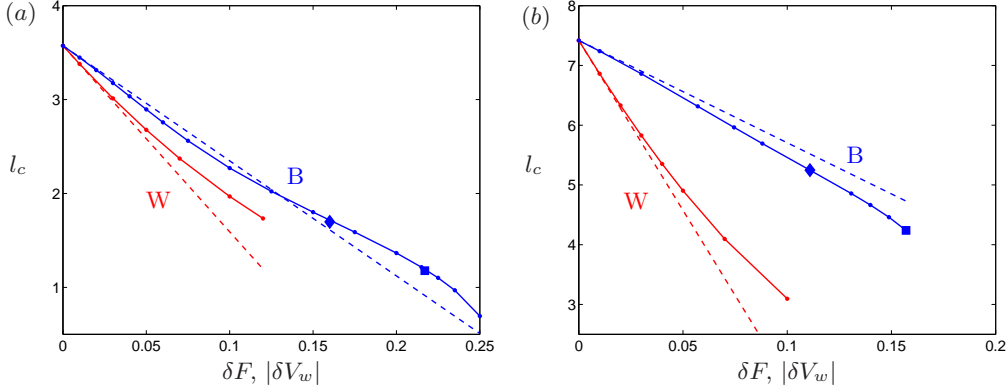


FIGURE 6. Variation of recirculation length with control amplitude (δF in configuration B, $|\delta V_w|$ in configuration W). (a) $Re = 60$, (b) $Re = 120$. Dashed lines show predictions from sensitivity analysis, solid lines are non-linear results. Symbols correspond to control cylinders of diameter $d = 0.05D$ (\blacklozenge) and $d = 0.10D$ (\blacksquare).

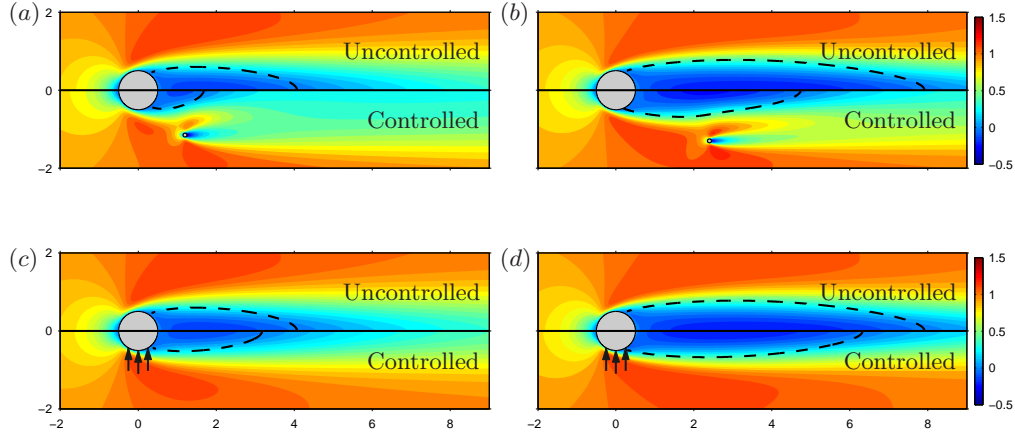


FIGURE 7. Streamwise velocity of uncontrolled and controlled steady-state base flow. Left: $Re = 60$; right: $Re = 120$. (a)-(b) configuration B: two control cylinders of diameter $d = 0.10D$ located at $(x^*, \pm y^*)$; (c)-(d) configuration W: suction of amplitude $\delta V_w = 0.05$ and 0.03 respectively.

- configuration W (“wall”; sketched in figure 5): vertical wall suction at the top and bottom sides, $\pi/3 \leq |\theta| \leq 2\pi/3$, with velocity δV_w .

Figure 6 shows the recirculation length variation δl_c for these configurations. In addition to predictions from sensitivity analysis (dashed lines), this figure shows non-linear results (solid lines) obtained by computing the actual controlled flow and its recirculation length for each configuration and each amplitude. In configuration B, the effect of control cylinders is modelled by the volume force (2.12); in configuration W, wall actuation is implemented as a velocity boundary condition with uniform profile. As expected, l_c is reduced, and the agreement between sensitivity analysis and non-linear results is excellent, with slopes matching at zero-amplitude, whereas non-linear effects appear for larger amplitudes. Non-linear simulations indicate that at $Re = 60$, controlling with two cylinders of diameter $d = 0.10D$ reduces l_c by more than 65%; a reduction of about 50%

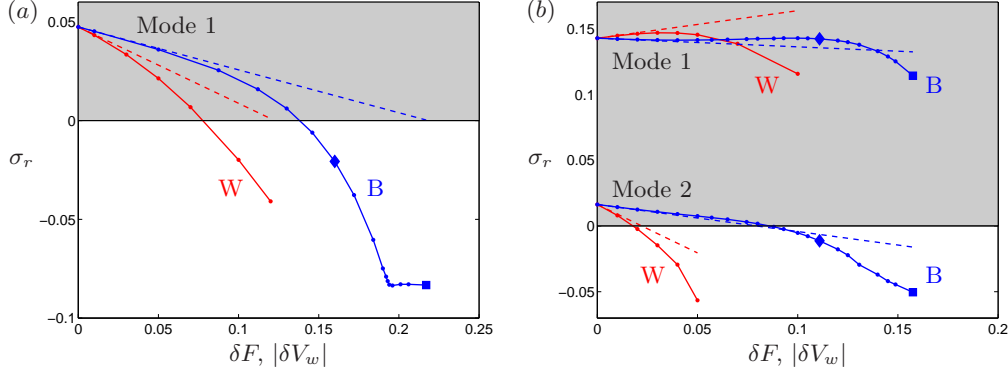


FIGURE 8. Variation of leading growth rates with control amplitude. (a) $Re = 60$, (b) $Re = 120$. Same notations as in figure 6.

is achieved with control cylinders of diameter $0.05D$, and with wall suction of intensity $\delta V_w = 0.12$. At $Re = 120$, control cylinders of diameter $d = 0.10D$ reduce l_c by more than 40%; a reduction of about 30% is achieved with control cylinders of diameter $0.05D$, and with wall suction of intensity $\delta V_w = 0.04$.

Figure 7 shows examples of controlled flows, illustrating how the recirculation region is shortened. In configuration B, control cylinders of diameter $d = 0.10D$ located at $(x^*, \pm y^*)$ make the flow deviate and accelerate. As a result, the streamwise velocity increases between the two control cylinders in a long region extending far downstream, and the reattachment point moves upstream. In configuration W, fluid is sucked at the cylinder sides, which brings fluid with high streamwise velocity closer to the wall region, which in turn makes the recirculation region shorter.

4. Effect on linear stability

In the previous section, sensitivity analysis was performed on the steady-state base flow. It provided information on the sensitivity of the recirculation length and allowed to design efficient control strategies. These results are relevant only if the controlled flow is stable. Indeed, when increasing the Reynolds number above its critical value Re_c , the cylinder flow becomes linearly unstable, with a Hopf bifurcation leading to unsteady vortex shedding; one should therefore investigate whether the flow is stabilised by the control. In the subcritical regime $Re < Re_c$, one should also check that the flow is not destabilised by the control.

At $Re = 60$, one pair of complex conjugate eigenvalues (“mode 1”) associated with the von Kármán street is unstable. Figure 8(a) shows the variation of the leading growth rate $\sigma_{1,r}$ with control amplitude. Both configurations B and W have a stabilising effect on the leading global mode. Full restabilisation is achieved with two control cylinders of diameter $d \simeq 0.04D$, or with wall suction of intensity $\delta V_w \simeq 0.08$. These control configurations do not destabilise other eigenmodes for any of the amplitudes tested.

A second pair of eigenvalues becomes unstable (“mode 2”) at $Re \simeq 110$ (Verma & Mittal 2011). Figure 8(b) shows the variation of $\sigma_{1,r}$ and $\sigma_{2,r}$ with control amplitude at $Re = 120$. Bulk forcing has a stabilising effect on both unstable modes, but does not achieve full restabilisation. The straightforward strategy which allowed to stabilise the flow at $Re = 60$, namely placing control cylinders where the sensitivity analysis predicts they have the largest reducing effect on l_c , is therefore not successful at $Re = 120$. Wall forc-

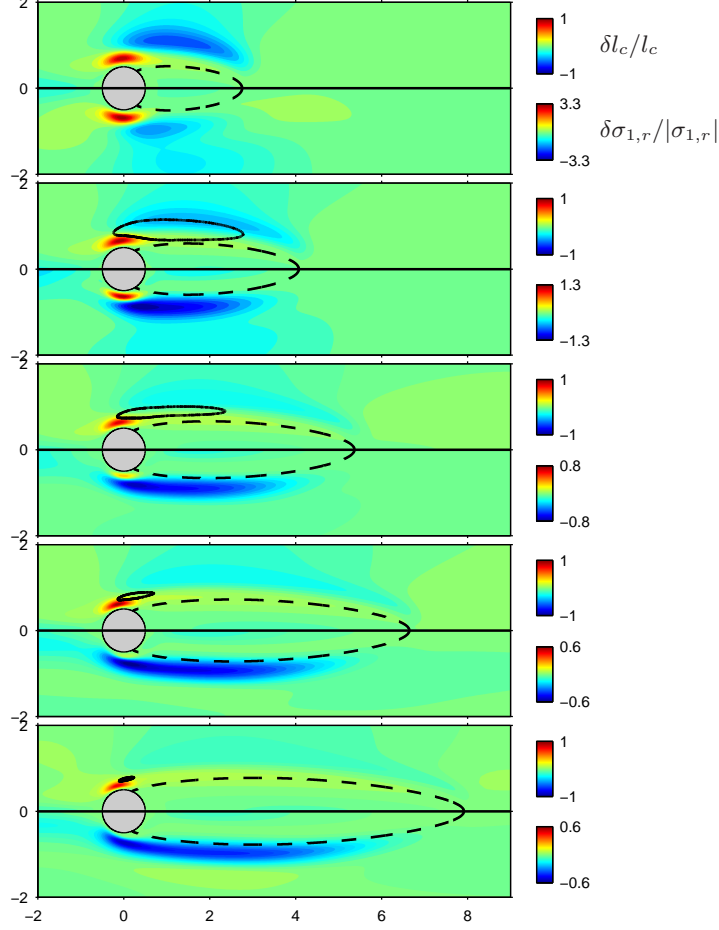


FIGURE 9. Normalized effect of a small control cylinder ($d = 0.10D$) on the recirculation length, $\delta l_c / l_c$ (upper half of each panel), and on the most unstable growth rate, $\delta \sigma_{1,r} / |\sigma_{1,r}|$ (lower half). From top to bottom: $Re = 40, 60, 80, 100, 120$. The stabilising contour $\delta \sigma_{1,r} / \sigma_{1,r} = -1/2$ for two symmetric control cylinders is reported on the upper half plots for $Re > Re_c$.

ing has a stabilising effect on mode 2 but a destabilising effect on mode 1 for reasonable control amplitude, and again the flow is not restabilised.

To investigate why control configurations which shorten the recirculation region also have a stabilising effect at $Re = 60$, but not at $Re = 120$, it is interesting to consider the sensitivity of the leading eigenvalue. Figure 9 compares the effect of a small control cylinder ($d = 0.10D$) on l_c , calculated from (2.13) and already shown in figure 3(b), and its effect on $\sigma_{1,r}$, calculated from (2.14). At $Re = 40$ and 60 , δl_c and $\delta \sigma_{1,r}$ have very similar spatial structures. This means that a small control cylinder, when located where it reduces the recirculation length, almost always has a stabilising effect on the most unstable global mode. However, this similarity gradually disappears as Re increases. Regions where δl_c and $\delta \sigma_{1,r}$ have opposite signs grow in size, and at $Re = 120$ they extend along the whole shear layers, both inside and outside the recirculation region. To ease comparison, the contour where two symmetric control cylinders ($d = 0.10D$) render mode 1 just neutrally stable (i.e. where $\delta \sigma_{1,r} = -\sigma_{1,r}/2$) is reported on the map of δl_c . At

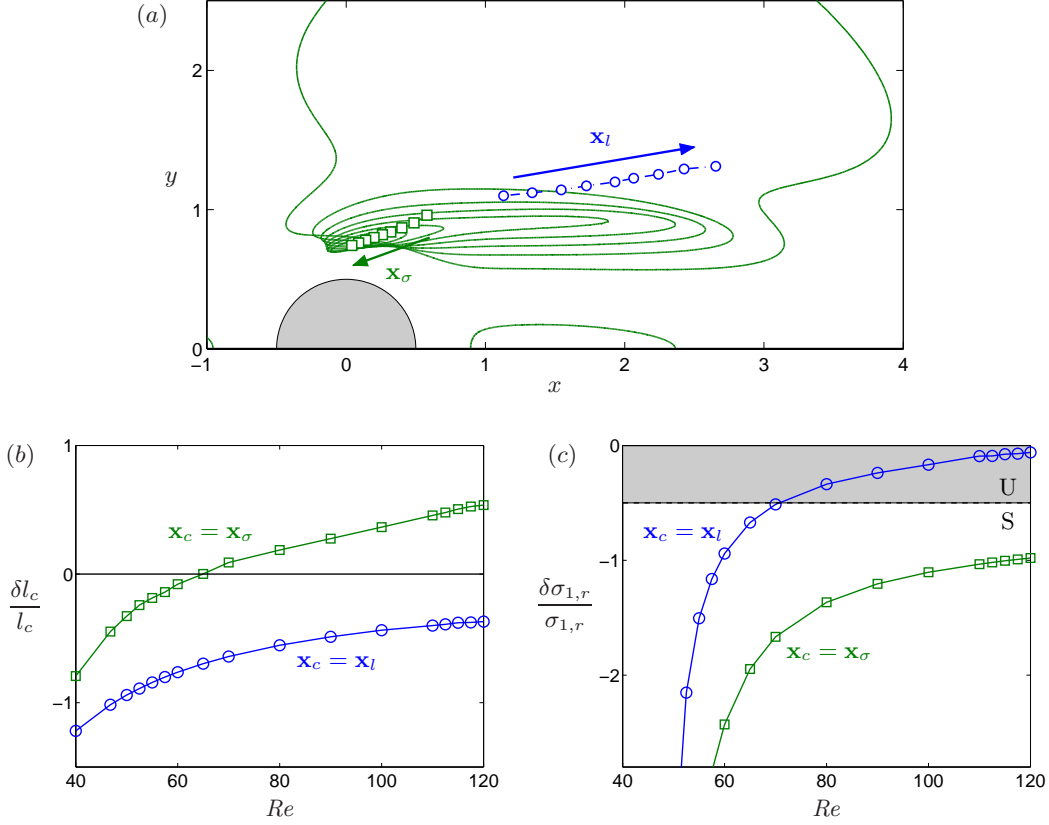


FIGURE 10. (a) Locus of \mathbf{x}_l (where $-\delta l_c/l_c$ is maximal) and \mathbf{x}_σ (where $-\delta \sigma_{1,r}/|\sigma_{1,r}|$ is maximal) for a small control cylinder ($d = 0.10D$) at $Re = 40, 50 \dots 120$. Solid lines show stabilising regions $\delta \sigma_{1,r}/\sigma_{1,r} \leq -1/2$ for two symmetric control cylinders at $Re = 50, 60 \dots 120$. (b)-(c) Effect of two symmetric control cylinders ($d = 0.10D$) located at \mathbf{x}_l or at \mathbf{x}_σ : sensitivity analysis prediction for the normalized variation of (a) recirculation length, (b) growth rate of mode 1. The unshaded area corresponds to restabilisation of mode 1.

$Re = 60$ this stabilising region overlaps with the region of recirculation length reduction, but as Re increases it moves upstream towards the region of recirculation length increase. In other words, control cylinders located where they reduce l_c are efficient in stabilising mode 1 at low Re , but gradually lose this ability at higher Re . In the latter regime, one may wonder if increasing the recirculation length is not a better way to stabilise the flow. It must be pointed out that the stabilising region shrinks as Re increases anyway, consistent with observations from Strykowski & Sreenivasan (1990) and Marquet *et al.* (2008).

All stabilising contours for $Re \geq 50$ are gathered in figure 10(a). The characteristic shrinking of the stabilising region is confirmed. Also shown for $Re \geq 40$ are the locus of two particular points: \mathbf{x}_l , where a control cylinder yields the largest l_c reduction, and \mathbf{x}_σ , where a control cylinder has the maximal stabilising effect on mode 1. It can clearly be observed that \mathbf{x}_l and \mathbf{x}_σ move in opposite directions, with \mathbf{x}_l eventually going outside the stabilising region. Quantitative values of δl_c and $\delta \sigma_{1,r}$ are given in figures 10(b)-(c). With control cylinders located at \mathbf{x}_l , a recirculation length reduction of more than 35% can be achieved for any $Re \leq 120$. (With $\mathbf{x}_c = \mathbf{x}_\sigma$, the reduction is of course not as large, and the recirculation length actually increases when $Re \gtrsim 65$.) As stressed previously, however, the corresponding controlled steady-state base flow will be observed only if stable, i.e. if

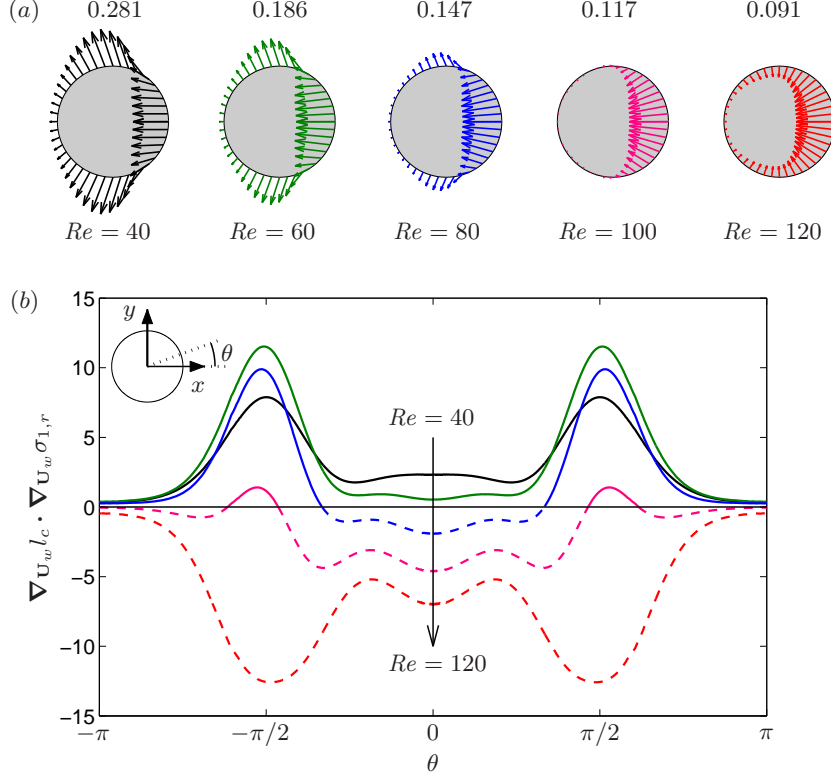


FIGURE 11. (a) Sensitivity of the leading growth rate to wall actuation $\nabla \mathbf{U}_w \sigma_{1,r}$. Flow is from left to right. Numbers correspond to the L^∞ norm of $\nabla \mathbf{U}_w \sigma_{1,r} / \sigma_{1,r}$. (b) Pointwise inner product of $\nabla \mathbf{U}_w l_c$ (figure 4) and $\nabla \mathbf{U}_w \sigma_{1,r}$ along the cylinder wall.

$\delta \sigma_{1,r} / \sigma_{1,r} \leq 1/2$ (region labelled S). This is the case if $\mathbf{x}_c = \mathbf{x}_\sigma$ (at least for $Re \leq 120$). But when control cylinders are located at \mathbf{x}_l , although they do have a stabilising effect ($\delta \sigma_{1,r} \leq 0$), the latter is too small to restabilise the flow when $Re \gtrsim 70$.

The same phenomenon is observed with sensitivity to wall forcing. Figure 11(a) shows the sensitivity of the most unstable growth rate to wall forcing. It compares very well with the results at $Re = 60$ of Marquet & Sipp (2010, figure 3 therein). Unlike the sensitivity of recirculation length (figure 4) which keeps more or less the same structure at all Reynolds numbers, $\nabla \mathbf{U}_w \sigma_{1,r}$ varies substantially. In particular, at the top and bottom sides of the cylinder, wall-normal suction changes from largely stabilising to slightly destabilising. This translates into $\nabla \mathbf{U}_w l_c$ and $\nabla \mathbf{U}_w \sigma_{1,r}$ being very similar at $Re = 40$ but very different at $Re = 120$. Accordingly, the pointwise inner product of these two fields shown in figure 11(b) is positive everywhere on the cylinder wall at $Re = 40$ but negative everywhere at $Re = 120$. As a result, control configurations which shorten the recirculation length necessarily have a stabilising effect on mode 1 at low Reynolds numbers and a destabilising effect at higher Reynolds numbers. It should be noted that at $Re = 100$, it is still possible to reduce l_c and at the same time have a stabilising effect on mode 1 by using wall suction in a narrow region near $\theta = \pm \pi/2$.

5. Conclusions

In this study, the sensitivity of recirculation length to steady forcing was derived analytically using a variational technique. Linear sensitivity analysis was applied to the two-dimensional steady flow past a circular cylinder for both subcritical and supercritical Reynolds numbers $40 \leq Re \leq 120$. Regions of largest sensitivity were identified: l_c increases the most when small control cylinders are located close to the top and bottom sides of the main cylinder, and decreases the most when control cylinders are located farther downstream, outside the shear layers; regarding wall forcing, l_c is most sensitive to normal blowing/suction at the sides of the cylinder. Validation against full non-linear Navier-Stokes calculations showed excellent agreement for small-amplitude control.

Using linear stability analysis, it was observed that control configurations which reduce l_c also have a stabilising effect on the most unstable eigenmode close to Re_c , both for bulk forcing and wall forcing. This property gradually disappears as Re is increased. This is explained by the spatial structures of the sensitivities of l_c and $\sigma_{1,r}$, which are very similar at low Re , but increasingly decorrelated at larger Re . Therefore, reducing the base flow recirculation length is an appropriate control strategy to restabilise the flow at moderate Reynolds numbers. At larger Re , aiming for an increase of l_c is actually more efficient. In any case, one should keep in mind that results concerning l_c reduction and obtained from a sensitivity analysis performed on the steady-state base flow are relevant only when the controlled flow is stable.

To better control the flow in the supercritical regime, the sensitivity of the mean flow recirculation length should be considered. Not only would a method allowing to control the mean l_c be interesting in itself, but targeting this important parameter of the mean state (Zielinska *et al.* 1997; Thiria & Wesfreid 2007) could also help stabilise the flow. This work is in progress, but the difficulty of such an approach lies in that the mean flow and the fluctuations are non-linearly coupled, which prevents the derivation of a simple expression for the sensitivity of the mean recirculation length. It would help, though, to determine whether the mean flow recirculation length in separated flows should be reduced or increased in order to mitigate the instability. This extension of the sensitivity analysis to the mean recirculation length could also include the effect of periodic excitation, a control strategy often used in turbulent flows (Greenblatt & Wygnanski 2000; Glezer & Amitay 2002) and which can be interpreted as a mean-flow correction (Sipp 2012).

This study confirms the high versatility of Lagrangian-based variational techniques, which allow to compute the sensitivity of a great variety of quantities of interest in fluid flows. The recirculation length appears as a simple and relevant macroscopic parameter in separated flows, and can be targeted to design original control strategies. In the case of the cylinder flow, the fact that l_c is a good proxy for flow stabilisation only up to a certain Reynolds number might be specific to the absolute nature of the instability in bluff body wakes (Monkewitz 1988). These flows are typical examples of “oscillators”, and are appropriately described by global linear stability analysis (Chomaz 2005). On the other hand, convectively unstable flows (or “noise amplifiers”), such as separated boundary layers, usually exhibit large optimal transient growth (maximal energy amplification of an initial perturbation) and large optimal gain (maximal energy amplification from harmonic forcing to asymptotic response for a globally stable flow) as a result of the non-normality of the linearized Navier-Stokes operator (Chomaz 2005). One may wonder whether the recirculation length is more directly and strongly related to instability in such convectively unstable flows, and whether sensitivity analysis applied to l_c would be

efficient over a broader range of Reynolds numbers. The next step of this work is the application of a similar control strategy in wall-bounded separated boundary layer flows.

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